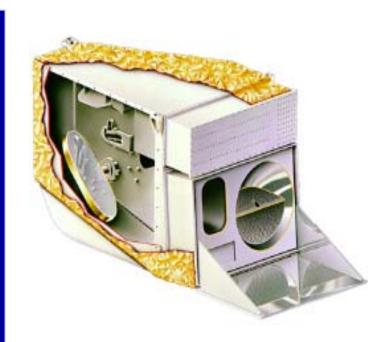


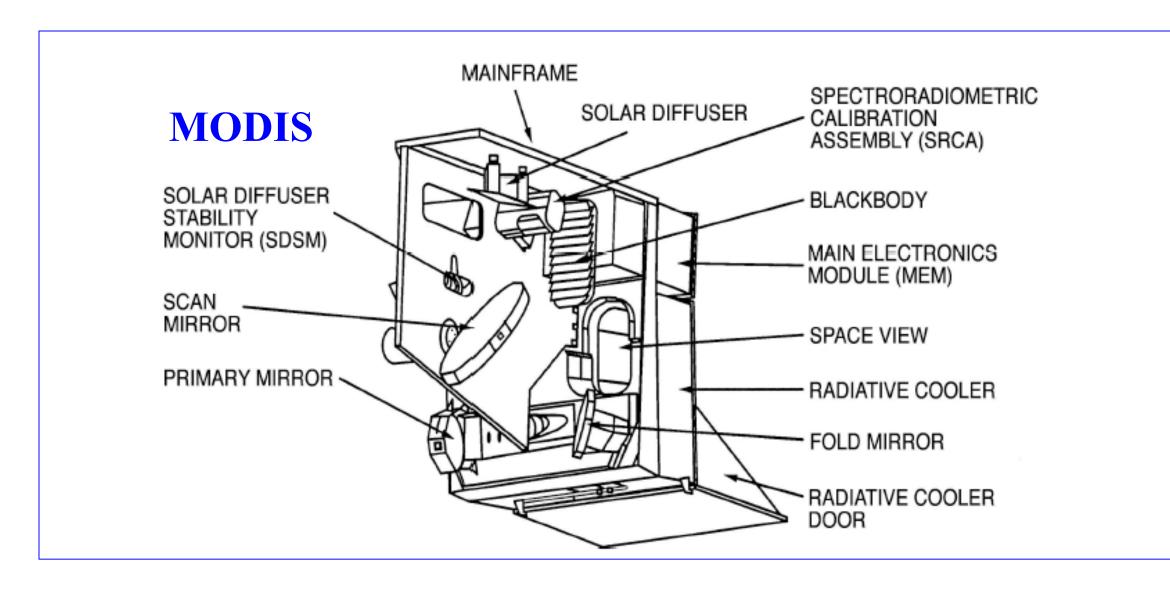
# MODIS Thermal Emissive Bands On-orbit Calibration



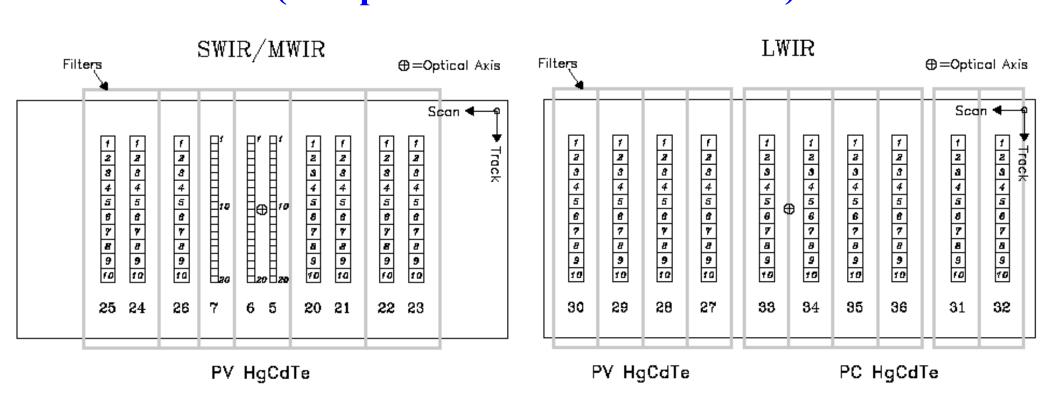
## MODIS Characterization Support Team

#### Introduction

- The MODerate Resolution Imaging Spectroradiometer (MODIS) is a key instrument for NASA's Earth Observing System (EOS). It is onboard both Terra and Aqua spacecrafts. The Terra spacecraft was launched on December 18, 1999 and the Aqua spacecraft was launched on May 4, 2002.
- MODIS has 36 spectral bands ranging from 0.42 to 14.5 microns, located on four focal plane assemblies (FPAs), with spatial resolution (at nadir) of 250 m (bands 1-2), 500 m (bands 3-7) and 1000 m (bands 8-36). There are 16 thermal emissive bands (TEB), B20-25 and B27-36, located on SMIR and LWIR focal planes controlled at 83K on-orbit.
- MODIS 2-sided paddle wheel scan mirror provides a -55 to +55 degree scan of the Earth View (EV) covering a 10 km (at nadir) along track by 2330 km along scan swath.



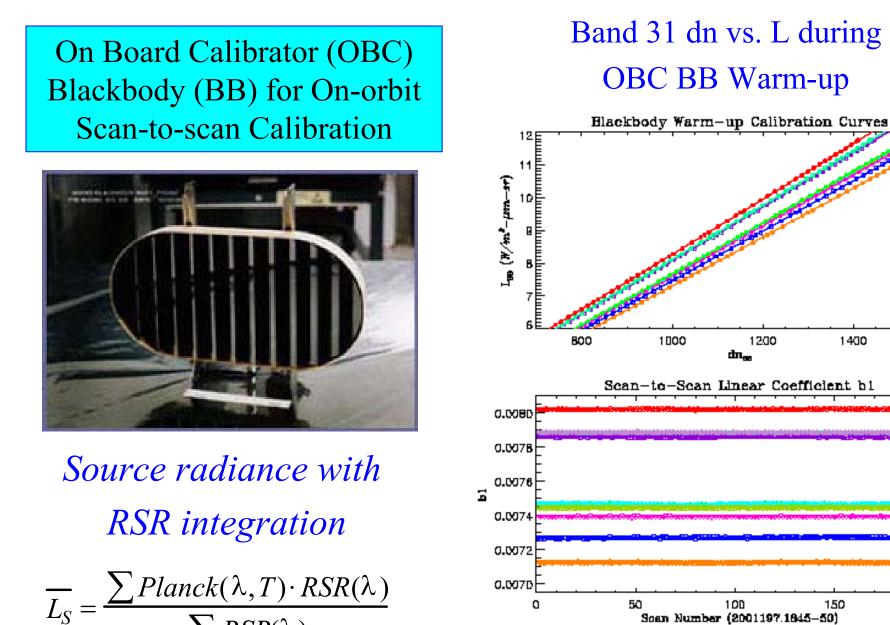
## SMIR and LWIR Cold Focal Plane Assemblies Layout (Temperature controlled at 83K)



#### **TEB Key Specifications**

TED ILLY Specifications							
Band	λ <sub>ctr</sub> (μm)	L <sub>typ</sub> Radiance (W/m²-μm-sr)	Scene Temperature at L <sub>typ</sub> (K)	Required NEdT (K)	On-orbit NEdT at L <sub>typ</sub> (Avg)	Radiometric Requirement at $L_{typ}$ (%) <sup>1</sup>	Radiometric Requirement at L <sub>typ</sub> (K)
20	3.79	0.45	300	0.05	0.03	0.75%	0.18
21	3.99	2.38	335	0.20	0.16	1%	0.31
22	3.97	0.67	300	0.07	0.03	1%	0.25
23	4.06	0.79	300	0.07	0.02	1%	0.25
24	4.47	0.17	250	0.25	0.16	1%	0.19
25	4.55	0.59	275	0.25	0.06	1%	0.24
27	6.77	1.16	240	0.25	0.10	1%	0.27
28	7.34	2.18	250	0.25	0.05	1%	0.32
29	8.52	9.58	300	0.05	0.02	1%	0.53
30	9.73	3.69	250	0.25	0.10	1%	0.42
31	11.01	9.55	300	0.05	0.03	0.5%	0.34
32	12.03	8.94	300	0.05	0.03	0.5%	0.37
33	13.36	4.52	260	0.25	0.13	1%	0.62
34	13.68	3.76	250	0.25	0.20	1%	0.59
35	13.91	3.11	240	0.25	0.23	1%	0.55
36	14.19	2.08	220	0.35	0.43	1%	0.47
$^{1}$ defined at $L_{ m typ}$ and between $\pm 45^{ m o}$ off nadir							

#### **TEB On-orbit Calibration Algorithm**



#### TEB EV Radiance Retrieval (TOA)

 $\sum RSR(\lambda)$ 

 $\overline{L}_{EV}(\theta) = \frac{1}{RVS_{EV}(\theta)} \cdot \left\{ \left( a_0(T_{inst}) + b_1 \cdot dn_{EV(\theta)} + a_2(T_{inst}) \cdot dn_{EV(\theta)}^2 \right) - \left( RVS_{SVS} - RVS_{EV}(\theta) \right) \cdot \overline{L}_{SM} \right\}$ 

#### Linear calibration coefficient b<sub>1</sub>

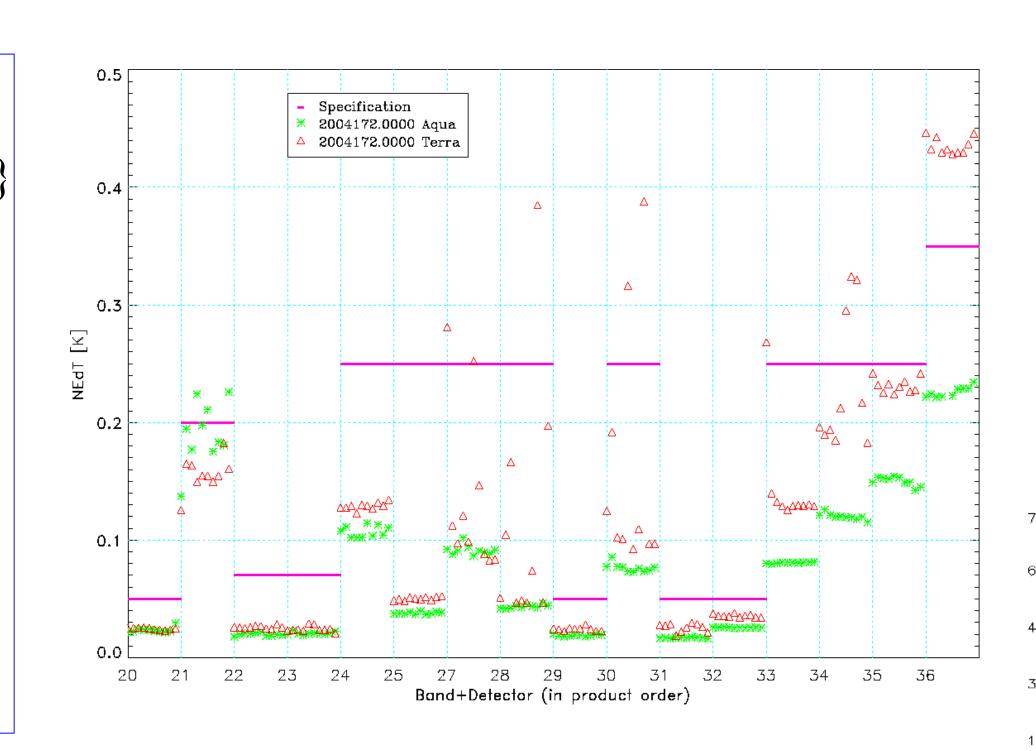
$$e_{1} = \frac{\varepsilon_{BB} \cdot \overline{L}_{BB} + (RVS_{SV} - 1)\overline{L}_{SM} + (1 - \varepsilon_{BB}) \cdot \varepsilon_{cav} \cdot \overline{L}_{cav} - a_{0} - a_{2} \cdot dn_{BB}^{2}}{dn_{BB}}$$

- $a_0(T_{inst})$  and  $a_2(T_{inst})$ : offset and the non-linear terms extracted from the
- pre-launch calibration and updated on-orbit from the OBC BB
- b<sub>1</sub>: the linear coefficient measured on-orbit from the OBC BB
- $RVS_{EVSV}$ : response versus scan angle at a given AOI •  $dn_{EV BB}$ : background subtracted instrument signal (digital number)
- L<sub>s</sub>: source radiance
- $\varepsilon_s$ : source emissivity

#### **Blackbody and Cavity Emissivity**

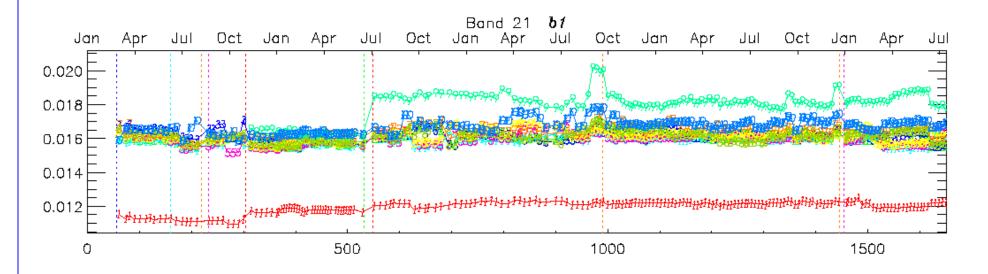
- Blackbody emissivity (ε) was characterized pre-launch. The Earth view radiance uncertainty caused by the error of the Blackbody emissivity is less than 0.2% at L typical for all TEB.
- Cavity emissivity was set to 0.85 carrying uncertainty of 0.05 for all TEB prior to launch. It results less than 0.05% uncertainty in radiance retrieval at L typical for all TEB.

### Terra and Aqua NEdT Comparison



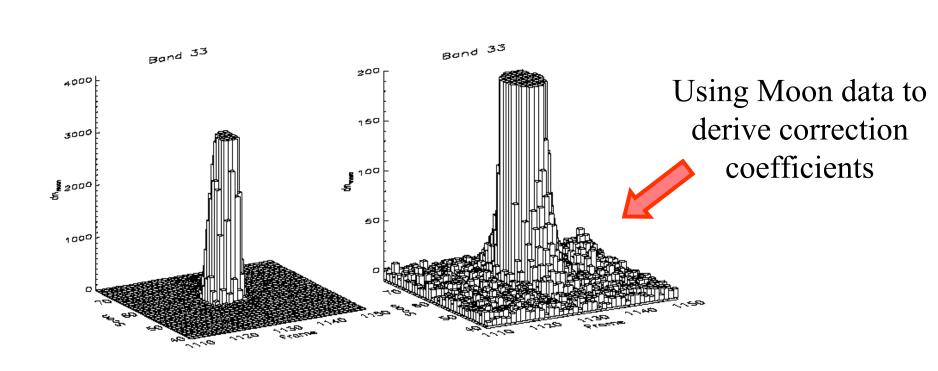
## **Band 21 (Fire Detection) Calibration**

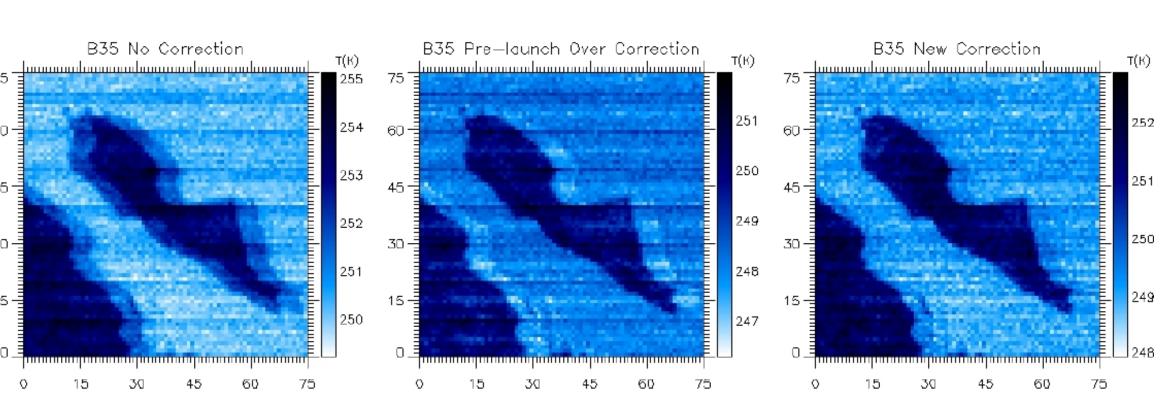
• On-orbit OBC Blackbody warm-up cycle limits to 270-315K. Four years of Terra MODIS regular Blackbody b<sub>1</sub> at 290K shows stable trends, except for one detector.



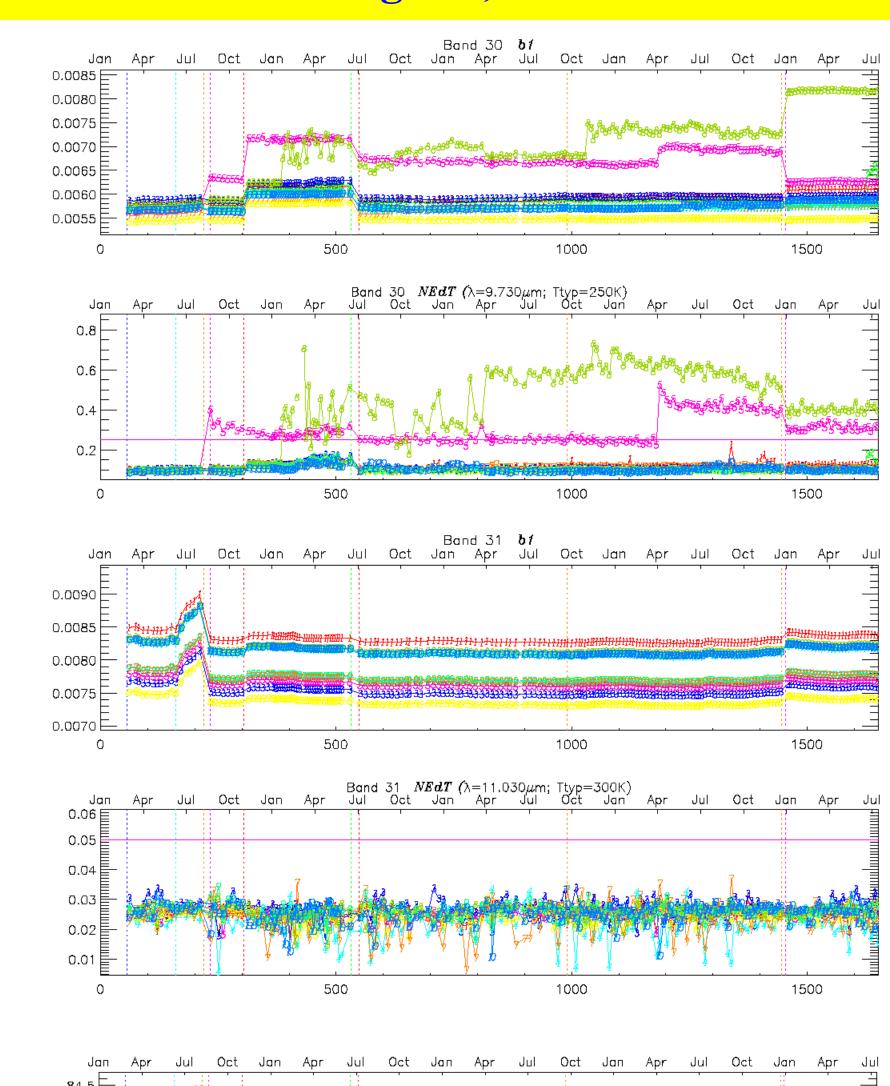
## LWIR PC Bands 11 µm Optical Leak - from B31 Filter into B32-36 Substrate

Band 33 detector 5 response over the Moon



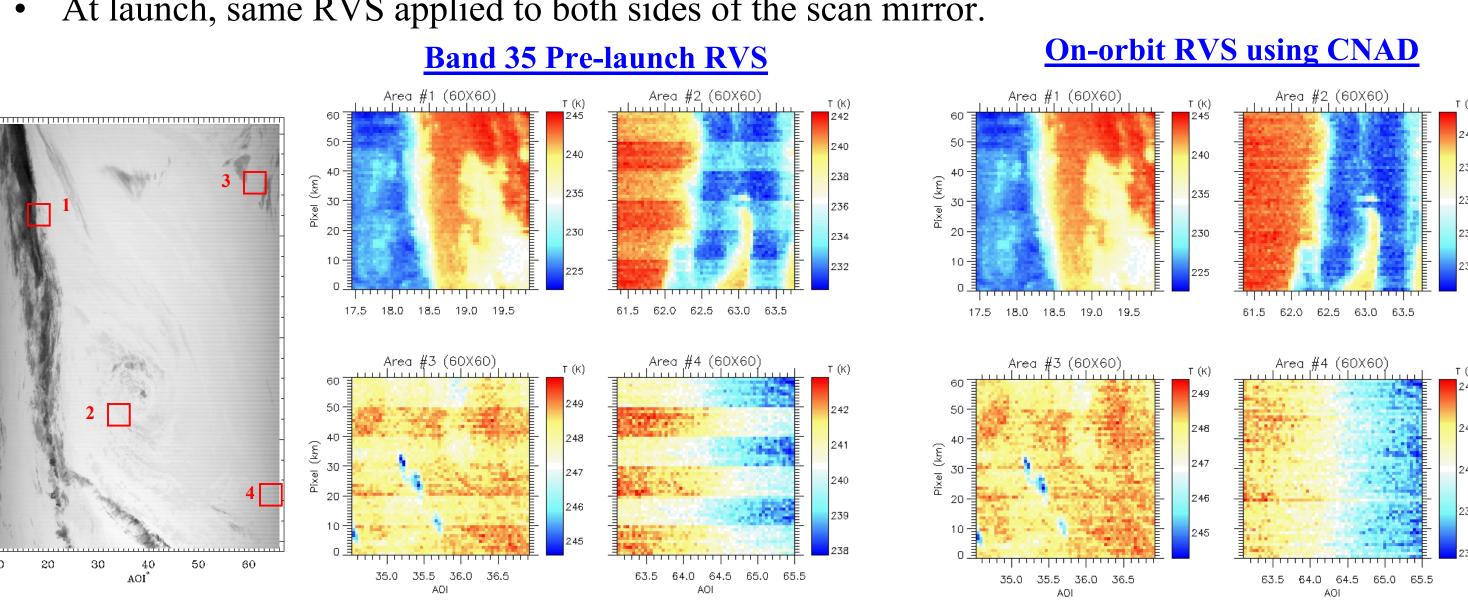


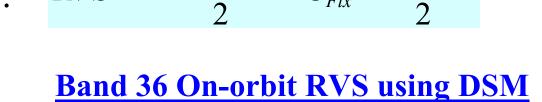
## On-orbit Trending: b1, NEdT and FP Temp.

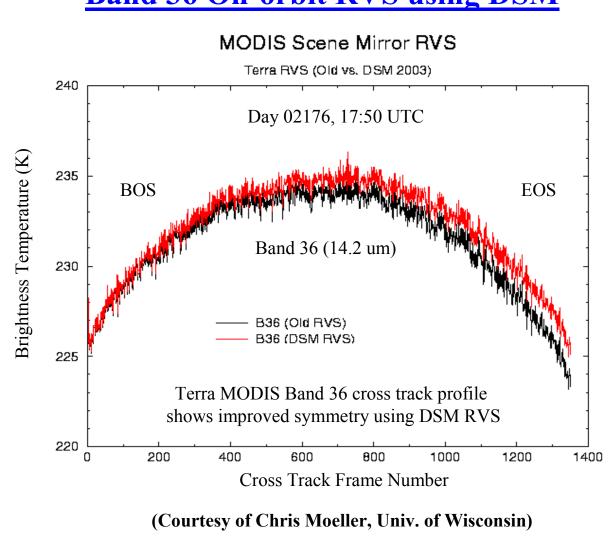


#### Response Versus Scan Angle (RVS) of the Scan Mirror

- No valid system level RVS measurements for TEB of Terra MODIS (PFM).
- Witness sample reflectance measurements (by NPL) and fixed optics parameters ( $C_{FIX}$ ) derived from the Aqua MODIS (FM1) system level measurements RVS are used to derive the PFM (Terra) RVS:
- At launch, same RVS applied to both sides of the scan mirror.







### **Challenging Issues**

- (1) Uncertainty and stability of band 21 at high temperatures.
- (2) Stripping and noisy detectors.